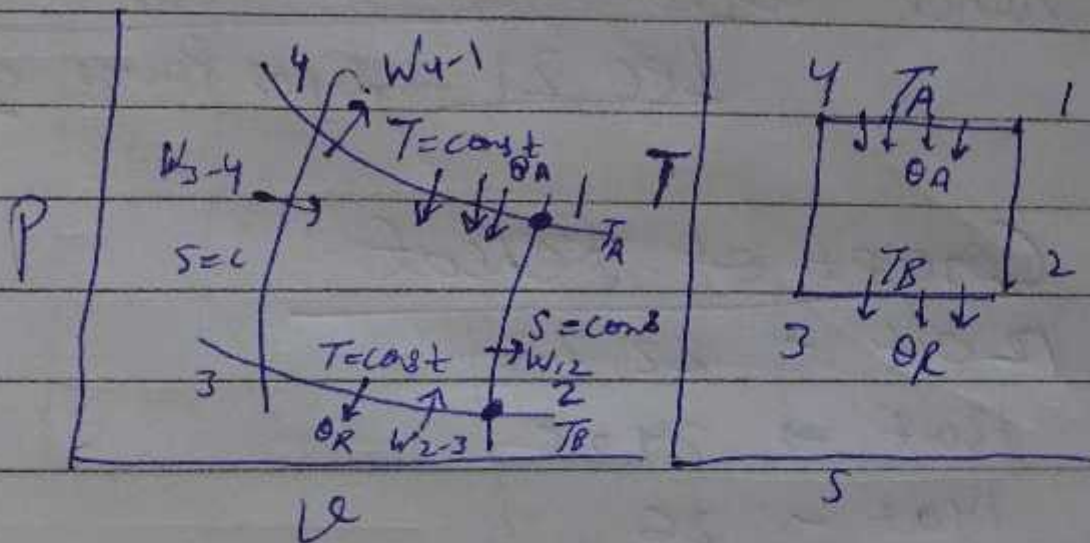


# Gas Power cycles I

Carnot cycle = Ideal Gas/Air  
Non flow process



Process	$\Delta U$	$W$ (Work done)	$Q$ (heat added)
1-2	$C_v(T_2 - T_1)$	$C_v(T_1 - T_2)$	0
2-3	0	$R T_B \ln(v_3/v_2)$	$R T_B \ln(v_3/v_2)$
3-4	$C_v(T_4 - T_3)$	$C_v(T_3 - T_4)$	0
4-1	0	$R T_A \ln(v_1/v_4)$	$R T_A \ln(v_1/v_4)$
	$\Sigma \Delta U = 0$	$\Sigma W =$	$\Sigma Q$

$$Q = \Delta U + W$$

$$T_1 = T_4 = T_A$$

$$T_2 = T_3 = T_B$$

$$W = \int P dv \text{ (non flow process)}$$

$$R T_B \ln \frac{v_3}{v_2}$$

08/11/2019

$$\eta = 1 - \frac{R T_B \ln(V_3/V_2)}{R T_A \ln(V_1/V_4)}$$

$$= 1 - \frac{T_B \ln(V_2/V_3)}{T_A \ln(V_1/V_4)}$$

$$P V^{\gamma} = \text{const} \quad \gamma = C_p/C_v$$

$$P V = R T$$

$$T V^{\gamma-1} = \text{const}$$

$$\frac{V_1}{V_2} = \left( \frac{T_1}{T_2} \right)^{1/1-\gamma}$$

$$\frac{V_4}{V_3} = \left( \frac{T_4}{T_3} \right)^{1/1-\gamma}$$

$$T_1 = T_4, \quad T_2 = T_3$$

$$\frac{V_1}{V_2} = \frac{V_4}{V_3}$$

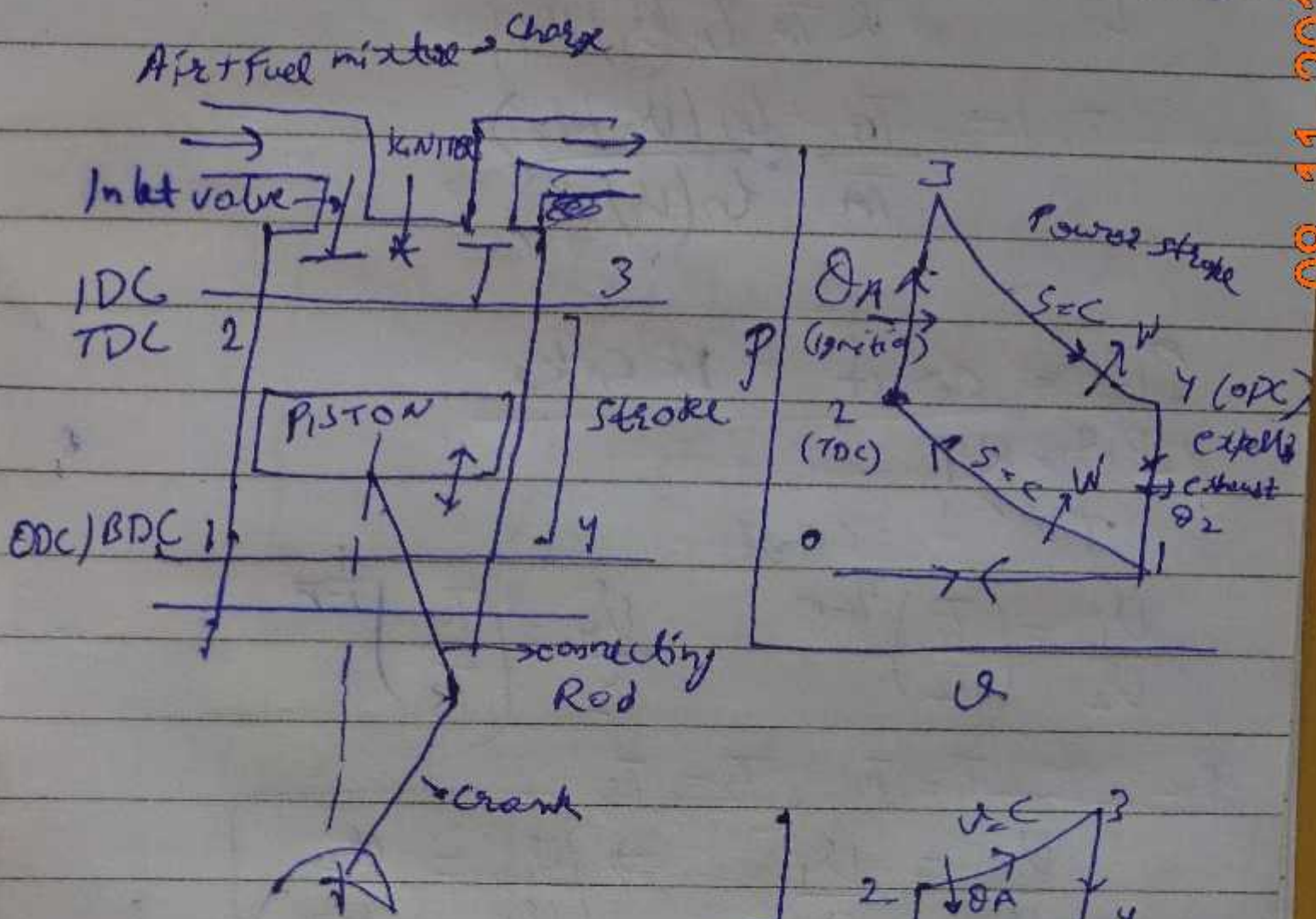
$$\Rightarrow \boxed{\frac{V_1}{V_4} = \frac{V_2}{V_3}}$$

$$\eta = 1 - \frac{T_B}{T_A}$$

Two Areas under P-V & T-S curve are the same  $\oint Q = \oint W$

# Otto Cycle - Air standard

## Reciprocating engine - Spark Plug



- Inlet valve open
- Piston reaches IDC/TDC
- Piston moves down
- Gas expands  $V$  increase  $P$  decrease (suction)
- Air comes in. (charge) (0-1)
- Piston goes to ODC/BDC.
- Inlet Valve closes. (both close)
- TDC Piston Ascends up. (1-2)  $s = \text{const}$ .
- Charge compressed. (high P, T) [2nd stroke]
- (2-3) IGNITER introduced, burning starts. (high P, T) gas. (IDC)
- Pushes piston down  $\downarrow$  const V

3-4

Piston goes down (expanding products of combustion  
 (IDC to ODC Power stroke) [3rd stroke] [3-4] isentropic  
 $T, P$  reduced. (Piston at ODC/BDC)

4-1 Outlet valve opens (even before piston reaches ODC)  
 Some gases expelled exhaust duct (Piston at ODC)

1-0 Piston ascends upwards. [1-0]  
 Mass goes out [ $P = P_{atm}$ ]

Piston reaches (IDC/TDC) [initial condition]

Again piston moves down. (inducts fresh charge)  
 new cycle

1 cycle = 4 movements or 4 strokes. (IDC to ODC)

Petrol engine - spark ignition engine

1-2 ~~Air taken in~~ goes to IDC

3-4 goes to BDC.

Process	$\Delta U$	$W$	$Q$
1-2	$C_v(T_2 - T_1)$	$C_v(T_1 - T_2)$	0
2-3	$C_v(T_3 - T_2)$	0	$C_v(T_3 - T_2)$
3-4	$C_v(T_4 - T_3)$	$C_v(T_3 - T_4)$	0
4-1	$C_v(T_1 - T_4)$	0	$C_v(T_1 - T_4)$
	$\Sigma \Delta U = 0$	$\Sigma W =$	$\Sigma Q$

closed system / non flow process

$$Q = \Delta U + W$$

$$\eta = 1 + \frac{C_v(T_1 - T_4)}{C_v(T_3 - T_2)}$$

$$= 1 - \frac{T_4 - T_1}{T_3 - T_2}$$

## Gas Power Cycle II

Volumetric compression ratio

$$r_v = V_1/V_2$$

$$\eta = \left[ 1 - \frac{T_4 - T_1}{T_3 - T_2} \right]$$

$$\frac{T_3}{T_2} = r_v^{r-1} = \left( \frac{V_1}{V_2} \right)^{r-1}$$

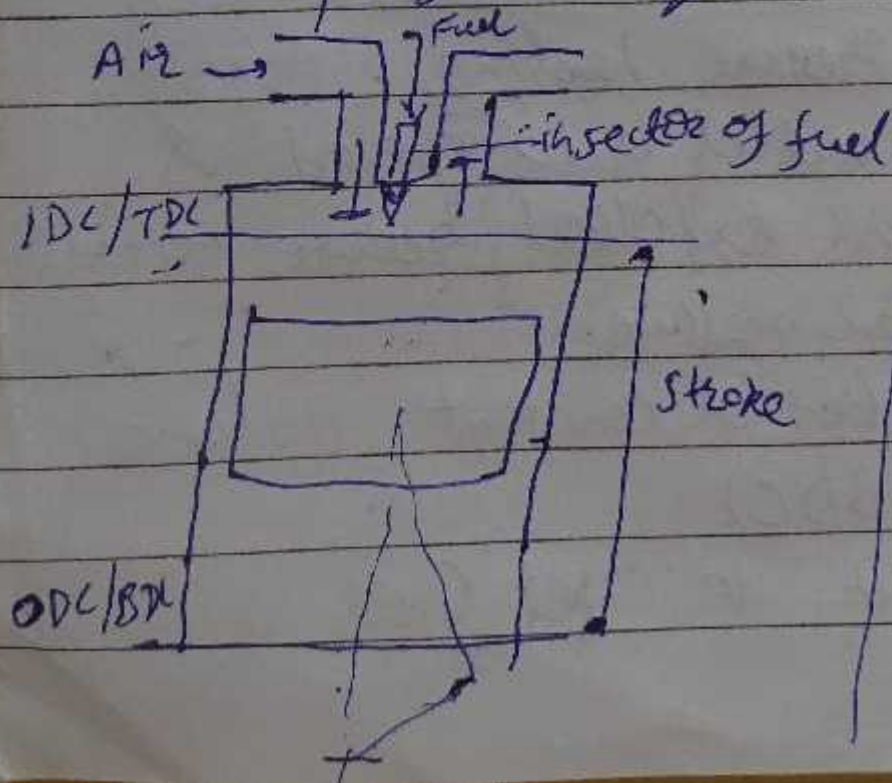
$$\frac{T_2}{T_1} = \frac{V_1}{V_2} = r_v^{r-1}$$

$$\eta = 1 - \frac{1}{r_v^{r-1}}$$

As we increase  $r_v$ ,  $\eta$  increases monotonically.

## Diesel cycle

Compression - ignition



Inlet stroke

IDC to moves down  
suction created.

Air moves in.

Piston reached BDC

Air is full, two valves  
closed. Piston ascends  
to IDC.

Pressure & Temp  
at end of compression  
is higher. Auto  
ignition temp.

Burning of liquid fuel. Mix vapor fuel & air. (molecular mixing)

Burning takes place in Gas phase

IGNITION Delay. time for auto ignition.

This takes place in carburetor of petrol engine.

Inside the cylinder, done in diesel engine

Injector directs a Jet of diesel

(spray of minute droplets - atomisation)

Surface Area increases  $\rightarrow$  rate of vaporization increases.

Immediate vaporization - burning. (droplet combustion) <sup>holys</sup>

Ignition starts heterogeneously at <sup>burning</sup> different points. Piston descends <sup>(rises)</sup> downwards by that time. (Power stroke)

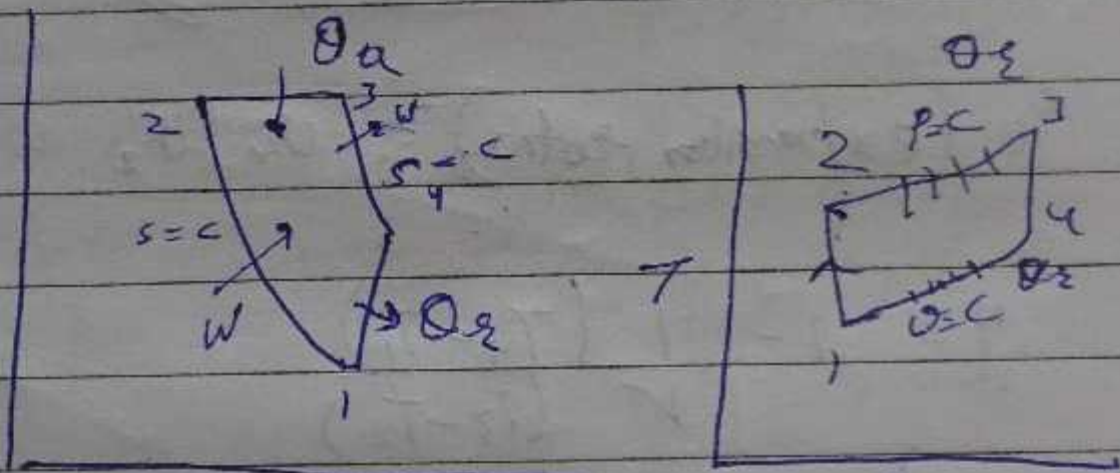
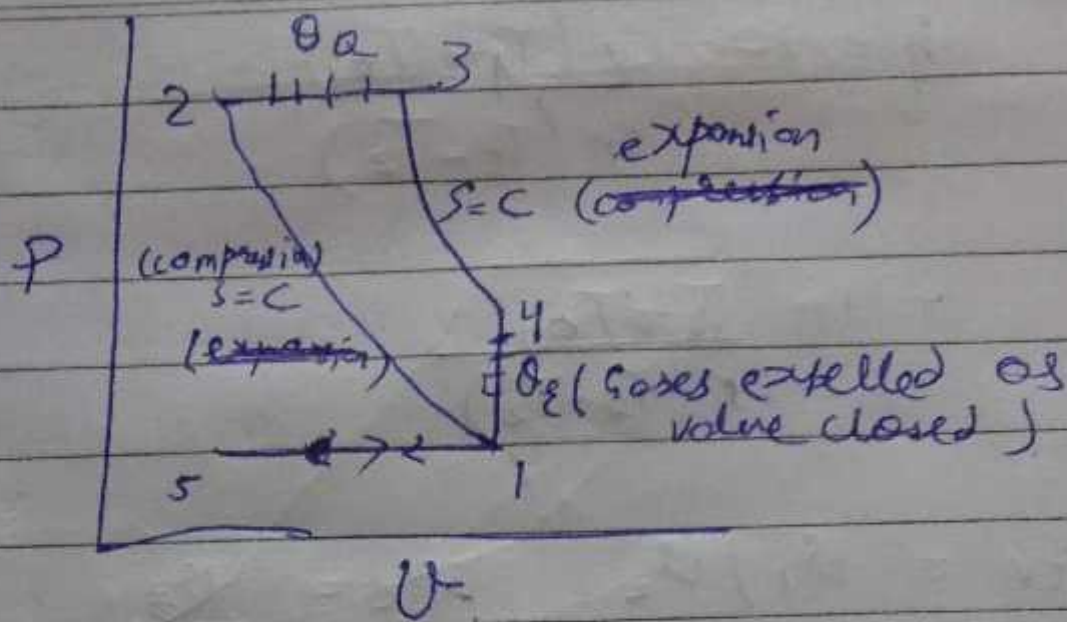
Constant Pressure heating.

Then ~~the~~ outlet valve opened & burning material expelled, pressure restored to atmospheric pressure.

Heat rejection at constant volume (Piston at BDC)

Piston moves up to expel Gases

# Diesel cycle



$P=c$  is less steep

Process	$\Delta U$	$w$	$Q$ (heat added)
1-2	$C_v(T_2 - T_1)$	$w(T_1 - T_2)$	0
2-3	$C_v(T_3 - T_2)$	$P_2(V_3 - V_2)$	$C_p(T_3 - T_2)$
3-4	$C_v(T_4 - T_3)$	$w(T_3 - T_4)$	0
4-1	$C_v(T_1 - T_4)$	0	$C_v(T_1 - T_4)$
	$\Delta U = 0$	$\Sigma w =$	$\Sigma Q$

Non flow process

$$Q = \Delta U + w$$

$$PdV = dU + PdV$$

$$\frac{d}{dn}(U + PV) = \text{const} + P$$



$$\eta = 1 - \frac{C_v (T_4 - T_1)}{C_p (T_3 - T_2)}$$

$$= 1 - \frac{T_4 - T_1}{\gamma (T_3 - T_2)}$$

$$\rho_v = \rho_1 / \rho_2 \quad \rho_c = \frac{\rho_3}{\rho_2}$$

(cut off ratio)

$$\rho_e (\text{expansion ratio}) = \rho_4 / \rho_3$$

$$\eta = 1 - \frac{1}{\gamma} \frac{(T_4 - T_1)}{(T_3 - T_2)}$$

$$\rho_e \rho_c = \rho_c \times \rho_e \quad (\rho_1 = \rho_4)$$

$$\frac{T_3}{T_4} = \left( \frac{\rho_4}{\rho_3} \right)^{\gamma-1} = (\rho_e)^{\gamma-1} \Rightarrow \frac{T_3}{T_4} = \left( \frac{\rho_v}{\rho_c} \right)^{\gamma-1}$$

$$\frac{T_2}{T_1} = \left( \frac{\rho_1}{\rho_2} \right)^{\gamma-1} = (\rho_v)^{\gamma-1}$$

$$\frac{T_3}{T_2} = \rho_c \left( \frac{\rho_3}{\rho_2} \right)$$

$$\eta = 1 - \frac{1}{\left( \rho_v \right)^{\gamma-1}} \cdot \frac{\rho_c^{\gamma-1} - 1}{\rho_c - 1}$$

Dieser

$$\epsilon_c > 1 \quad \frac{1}{\gamma} \left( \frac{\epsilon_c^\gamma - 1}{\epsilon_c - 1} \right) > 1$$

write  $\epsilon_c = 1 + x$   
 $\frac{1 + \gamma x + \dots}{\gamma x}$

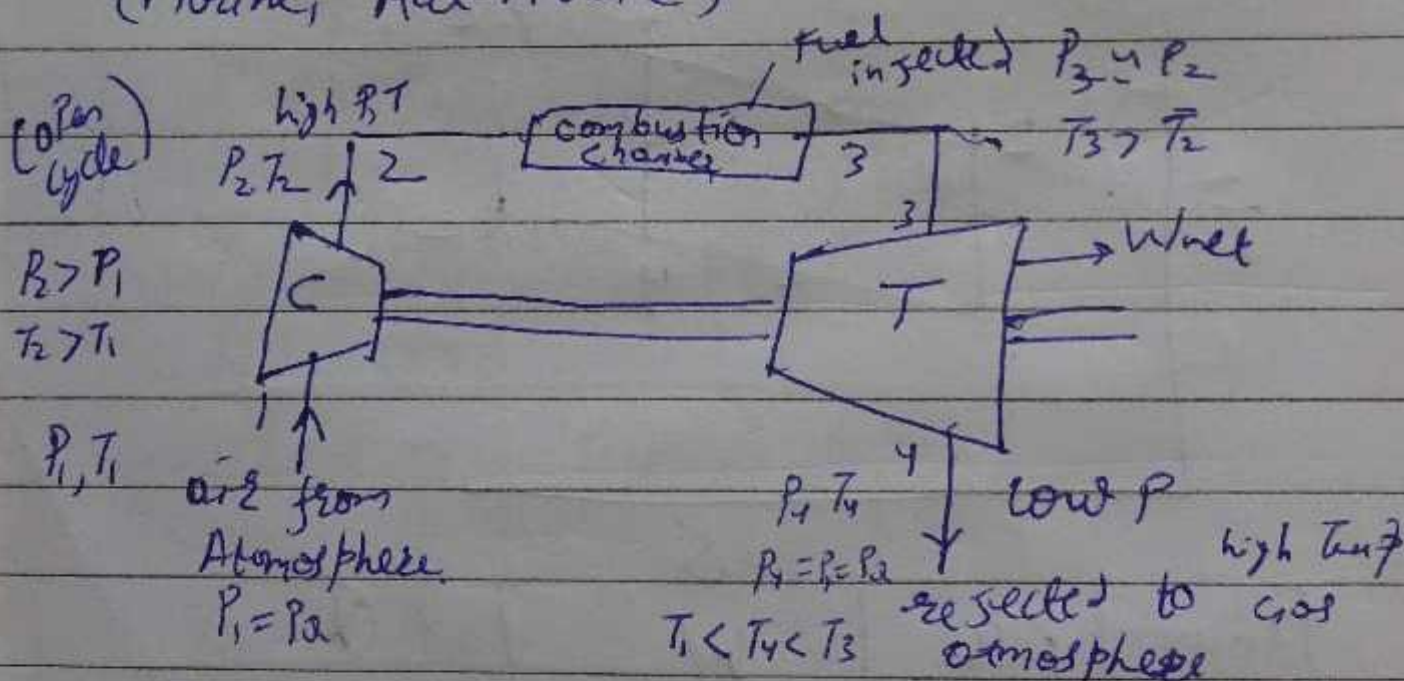
$$\eta_{otto} > \eta_{diesel}$$

For same value of  $\epsilon_c$

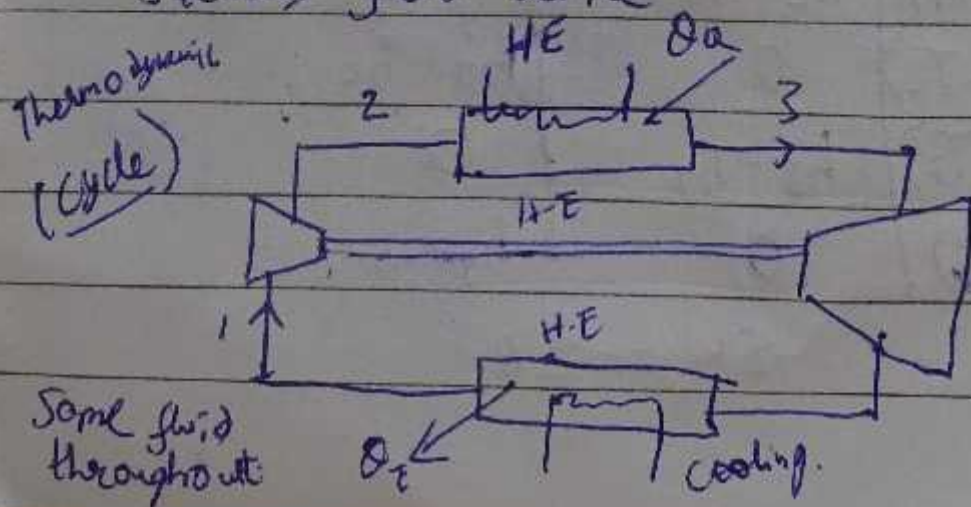
2  $\rightarrow$  represents burning process

Brayton cycle -  
 Gas turbine plant (Air craft engine)  
 Rocket engine

(Marine, Automobile)



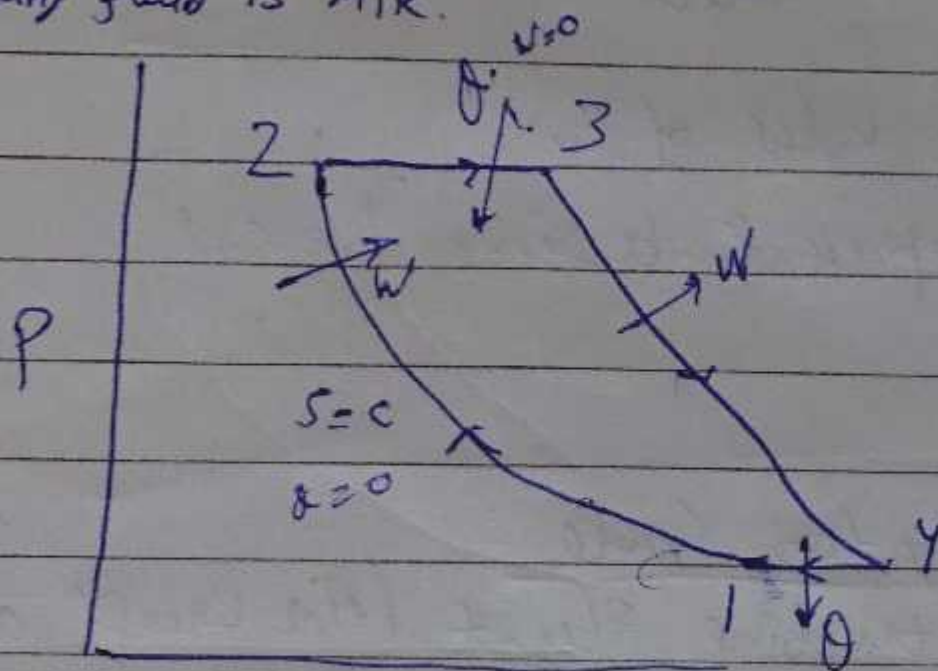
Steady flow device



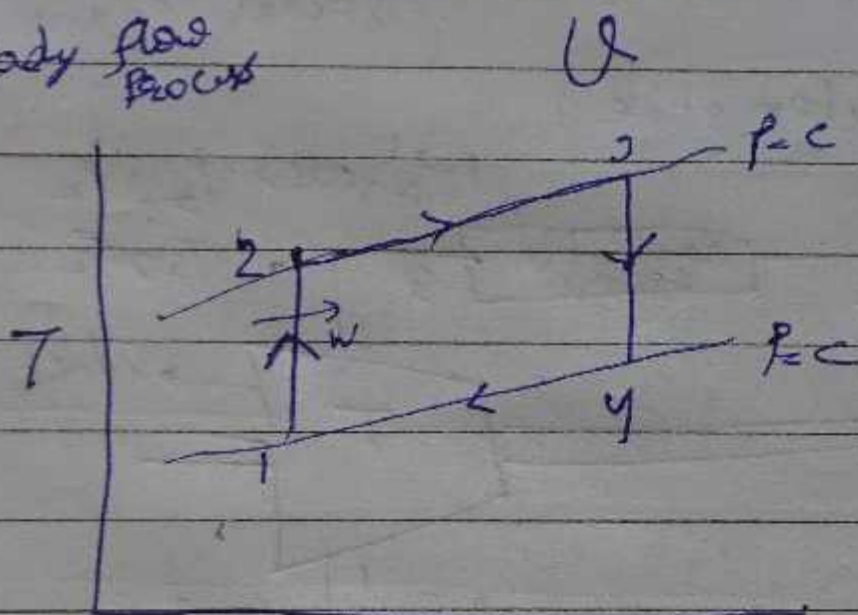
closed cycle  
 Gas turbine  
 plant  
 (classical)  
 reactor  
 [2-3]  
 fission.

# Brayton / Joule cycle

Working fluid is AIR.



Steady flow process



Process	$\Delta u$	$w$	$q$
1-2	$c_v(T_2 - T_1)$	$h_2 - h_1$	0
2-3	$c_v(T_3 - T_2)$	0	$h_3 - h_2$
3-4	$c_v(T_4 - T_3)$	$h_3 - h_4$	0
4-1	$c_v(T_1 - T_4)$	0	$h_4 - h_1$

$$\sum \Delta u = 0$$

$$\sum w = \sum q$$

$$\eta = \frac{W_{\text{net}}}{Q_{\text{Added}}} = 1 - \frac{Q_{\text{Rejected}}}{Q_{\text{Added}}}$$

$$= 1 - \left( \frac{h_4 - h_1}{h_3 - h_2} \right)$$

$$= 1 - \frac{C_p (T_4 - T_1)}{C_p (T_3 - T_2)} = 1 - \frac{(T_4 - T_1)}{T_3 - T_2}$$

$$P V^{\gamma} = \text{const}$$

$$P V = R T$$

$$P \left( \frac{R T}{P} \right)^{\gamma} = \text{const} \quad \frac{T}{P^{1/\gamma}} = \text{const}$$

$$\frac{T_3}{T_4} = \left( \frac{P_3}{P_4} \right)^{\gamma-1/\gamma} = \left( r_p \right)^{\gamma-1/\gamma}$$

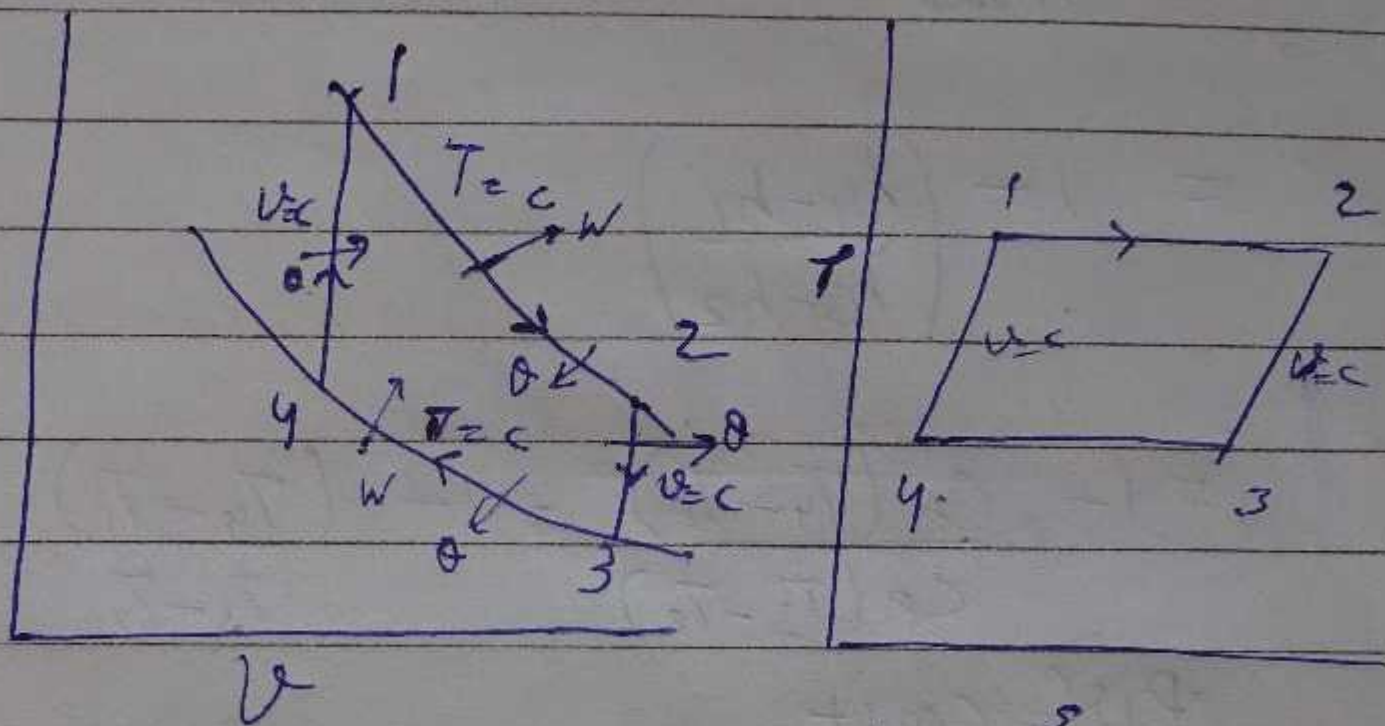
$$r_p = \frac{P_3}{P_4} = \text{Pressure Ratio}$$

$$\frac{T_2}{T_1} = \left( r_p \right)^{\gamma-1/\gamma}$$

$$\eta = 1 - \frac{1}{\left( r_p \right)^{\gamma-1/\gamma}}$$

# Stirling Cycle

Working system - AIR  
 non-flow  $\theta = \Delta U + W$



Process	$\Delta U$	$\theta$	$W$
1-2	0	$R T_1 \ln (v_2/v_1)$	$R T_1 \ln (v_2/v_1)$
2-3	$C_v (T_3 - T_2)$	$C_v (T_3 - T_2)$	0
3-4	0	$R T_3 \ln (v_4/v_3)$	$R T_3 \ln (v_4/v_3)$
4-1	$C_v (T_1 - T_4)$	$C_v (T_1 - T_4)$	0
	$\Sigma \Delta U = 0$	$\Sigma \theta =$	$\Sigma W$

$$\eta = 1 - \frac{\theta_{res}}{\theta_{add}} = 1 - \frac{C_v (T_2 - T_3) + R T_3 \ln (v_3/v_4)}{R T_1 \ln (v_2/v_1) + C_v (T_1 - T_4)}$$

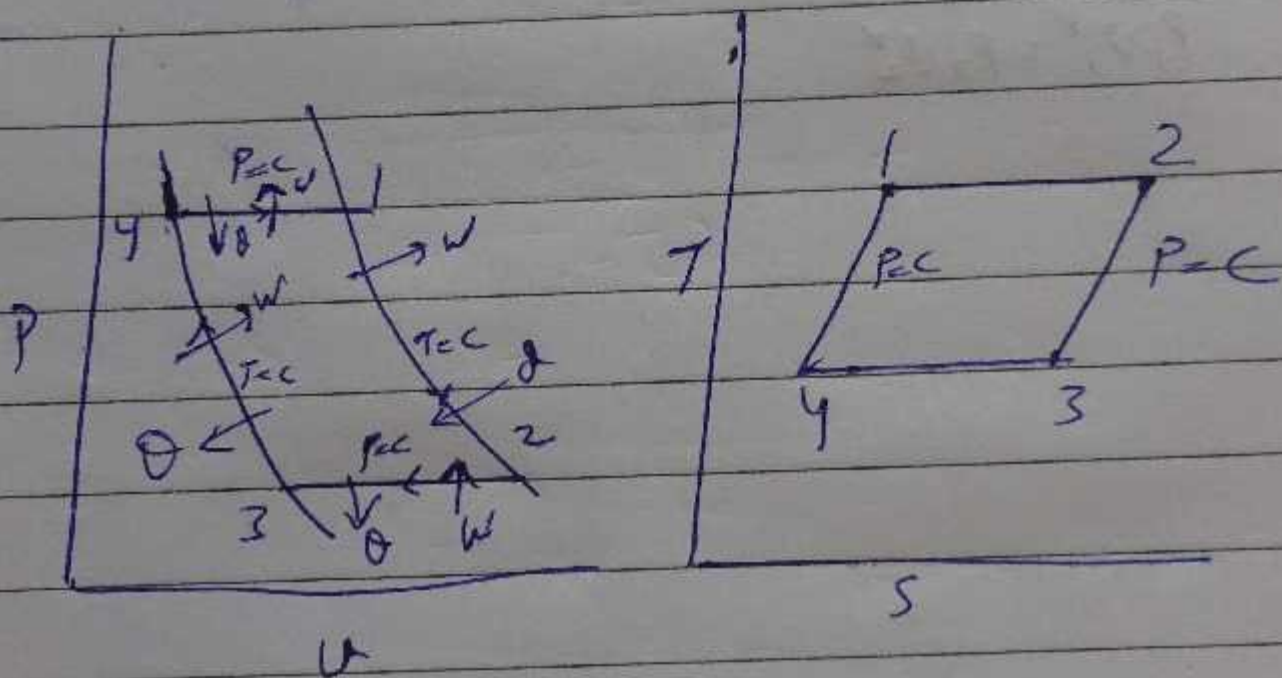
2-3 & 4-1 are counter.

Neglect:  $C_v (T_2 - T_3)$        $C_v (T_1 - T_4)$

$$\frac{Q_3}{Q_4} = \frac{Q_2}{Q_1}$$

$$\eta = 1 - \frac{T_3}{T_1} \rightarrow \text{Carnot efficiency}$$

Erisson cycle: working system Air  
non flow.  $Q = \Delta U + W$



Process	$\Delta U$	$Q$	$W$
1-2	0	$R T_1 \ln(v_2/v_1)$	$R T_1 \ln(v_2/v_1)$
2-3	$C_v(T_3 - T_2)$	$p_2(v_3 - v_2)$	$p_2(v_3 - v_2)$
3-4	0	$R T_3 \ln(v_4/v_3)$	$R T_3 \ln(v_4/v_3)$
4-1	$C_v(T_1 - T_4)$	$p_1(v_1 - v_4)$	$p_1(v_1 - v_4)$

$$\eta = 1 - \left[ \frac{R T_3 \ln(v_3/v_4) + C_p(T_2 - T_3)}{R T_1 \ln(v_2/v_1) + C_p(T_1 - T_4)} \right]$$

$$Q_{41} \rightarrow Q_{23}$$

only 1-2 & 3-4 :

$$\eta = 1 - \frac{T_3}{T_1}$$

$$P_3 v_3^r = P_4 v_4^r$$

$$P_1 v_1^r = P_2 v_2^r$$