

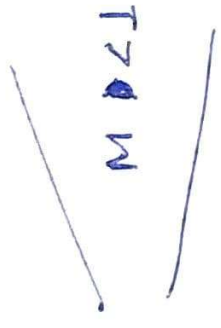
Pr. Diffuser.

$$dv = -ve. \text{ (always)}$$

i)  $M < 1$

$$\frac{dA}{A} = (-ve) (-ve) \Rightarrow +ve$$

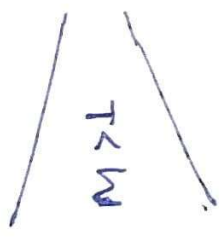
$$\frac{dA}{A} = +ve.$$



ii)  $M > 1$ ,

$$\frac{dA}{A} = (-ve) (+ve) \Rightarrow -ve$$

$$\frac{dA}{A} = -ve$$



Critical flow condition in nozzle!

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For. nozzle.

$$h_1 + \frac{v_1^2}{2} = h_2 + \frac{v_2^2}{2}$$

$$v_1 \ll v_2$$

$$h_1 = h_2 + \frac{v_2^2}{2}$$

$$h_1 - h_2 = \frac{v_2^2}{2} \Rightarrow v_2 = \sqrt{2(h_1 - h_2)}$$

$$v_2 = \sqrt{2c_p(T_1 - T_2)}$$

$$v_2 = \sqrt{2 \times c_p \cdot T_1 \left(1 - \frac{T_2}{T_1}\right)}$$

$$v_2 = \sqrt{2 \times c_p \cdot T_1 \left[1 - \left(\frac{p_2}{p_1}\right)^{\frac{\gamma-1}{\gamma}}\right]}$$

$$\frac{T_2}{T_1} = \left(\frac{p_2}{p_1}\right)^{\frac{\gamma-1}{\gamma}}$$

$$v_2 = \sqrt{2 \times c_p \cdot T_1 \left[1 - \left(\frac{p_2}{p_1}\right)^{\frac{\gamma-1}{\gamma}}\right]}$$

$$pV = mRT \Rightarrow p = \rho RT$$

$$\rho v^\gamma = \text{const} \quad \& \quad \frac{p}{\rho^\gamma} = \text{const} \Rightarrow$$

$$\frac{p_1}{\rho_1^\gamma} = \frac{p_2}{\rho_2^\gamma} \Rightarrow \rho_2 = \left(\frac{p_2}{p_1}\right)^{\frac{1}{\gamma}} \rho_1$$

$$\rho_1 = p_1 / RT_1$$

For. mass flow rate.

$$\dot{m} = \rho_2 A_2 v_2$$

$$\dot{m} = \rho_1 \times \left(\frac{p_2}{p_1}\right)^{\frac{1}{\gamma}} \times A_2 \times \sqrt{2 \times c_p \cdot T_1 \left[1 - \left(\frac{p_2}{p_1}\right)^{\frac{\gamma-1}{\gamma}}\right]}$$

$$\frac{\dot{m}}{A_2} = \rho_1 \times \left(\frac{p_2}{p_1}\right)^{\frac{1}{\gamma}} \times \sqrt{2 \times c_p \cdot T_1 \left[1 - \left(\frac{p_2}{p_1}\right)^{\frac{\gamma-1}{\gamma}}\right]}$$

$$\frac{\dot{m}}{A_2} = \rho_1 \sqrt{2 \times C_p \times T_1 \left[ \left( \frac{P_2}{P_1} \right)^{\frac{2}{\gamma}} - \left( \frac{P_2}{P_1} \right)^{\frac{\gamma+1}{\gamma}} \right]}$$

$$\frac{\gamma-1}{\gamma} + \frac{2}{\gamma} = \frac{\gamma-1}{\gamma} \quad (20)$$

value of  $\frac{P_2}{P_1}$  for max<sup>m</sup> flow rate.

$$\frac{d(\dot{m}/A_2)}{d(P_2/P_1)} = 0.$$

other quantities except the ratio  $\frac{P_2}{P_1}$  are constant.

$$\frac{2}{\gamma} \times \left( \frac{P_2}{P_1} \right)^{\frac{2}{\gamma}-1} - \frac{\gamma+1}{\gamma} \times \left( \frac{P_2}{P_1} \right)^{\frac{\gamma+1}{\gamma}-1} = 0$$

$$\frac{2}{\gamma} \left( \frac{P_2}{P_1} \right)^{\frac{2-\gamma}{\gamma}} = \frac{\gamma+1}{\gamma} \left( \frac{P_2}{P_1} \right)^{\frac{1}{\gamma}}$$

$$\frac{2}{\gamma+1} = \frac{\left( \frac{P_2}{P_1} \right)^{\frac{1}{\gamma}}}{\left( \frac{P_2}{P_1} \right)^{\frac{2-\gamma}{\gamma}}}$$

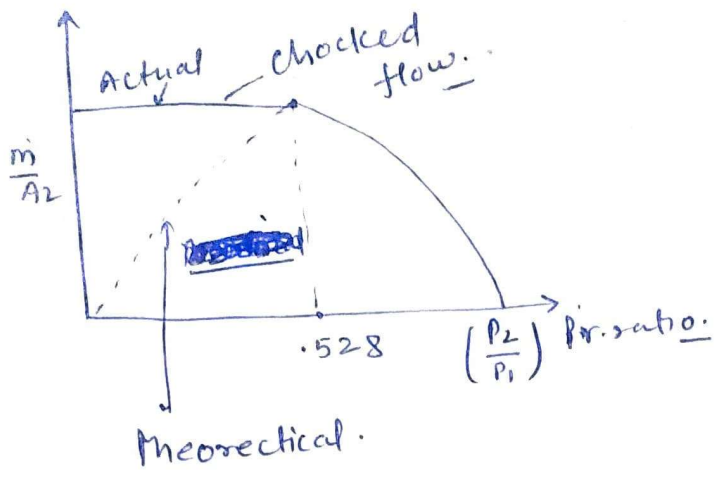
$$\frac{2}{\gamma+1} = \left( \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}}$$

$$\left( \frac{P_2}{P_1} \right) = \left( \frac{2}{\gamma+1} \right)^{\frac{\gamma}{\gamma-1}}$$

for  $\gamma = 1.4$ .

$$\boxed{\frac{P_2}{P_1} = 0.528}$$

↓  
critical pr. ratio.



Mass flow rate through convergent nozzle.

Fixed mass flow rate irrespective of pressure ratio is known as choked flow.