

18.1.22

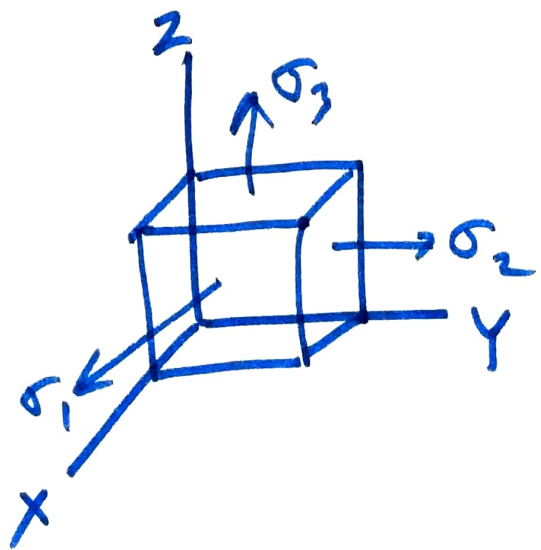
(17)

for a given set of axis = one principal axis
by changing the axis,
invariants should not change
they are not function of
choice of axis.

Principal plane
there will be

- only normal stress
- no shear stress

Metal deforms by shear stress

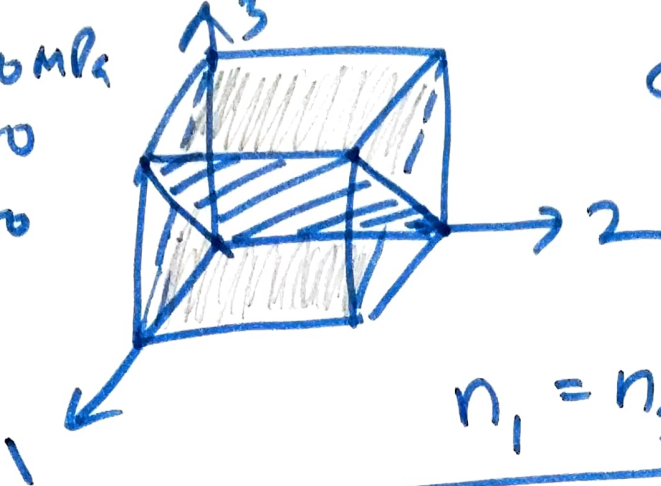


planes at an
angle 45°
will have
max^m
Shear
stress

$\sigma_2 = 500 \text{ MPa}$
 $\sigma_3 = 200$
 $\sigma_1 = 800$

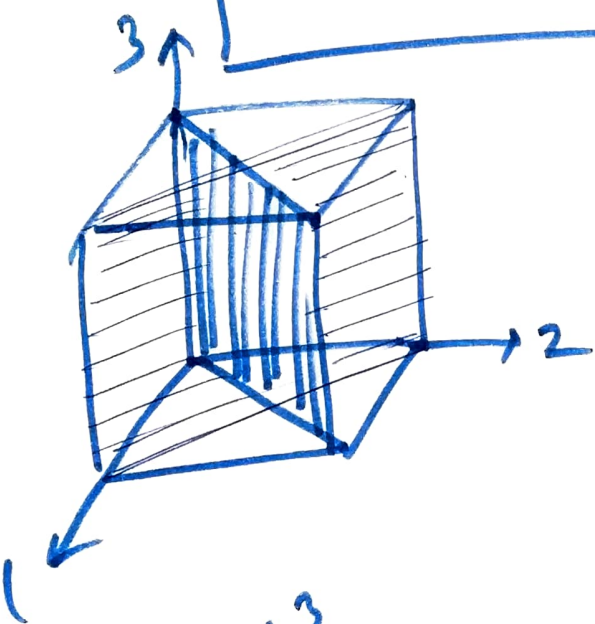
(18)

$\sigma_1 > \sigma_2 > \sigma_3$
 Largest \downarrow
 Smaller \downarrow



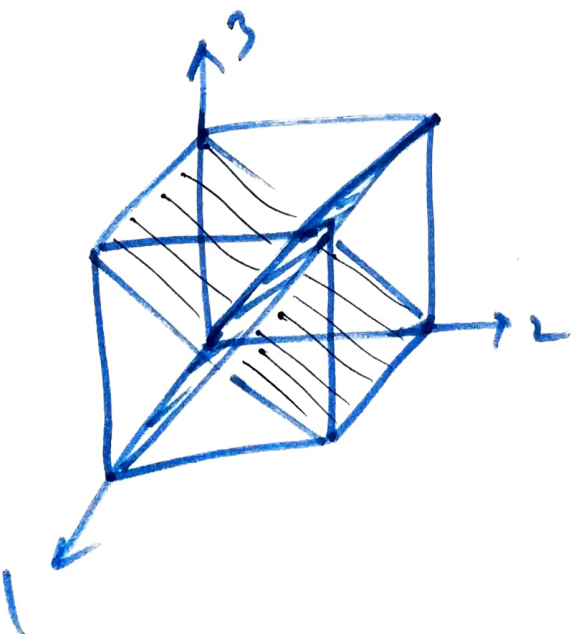
$$n_1 = n_3 = \pm \frac{1}{\sqrt{2}}$$

$$\tau_1 = \frac{\sigma_1 - \sigma_3}{2} \quad \text{max}^m$$



$$n_1 = n_2 = \pm \frac{1}{\sqrt{2}}$$

$$\tau_2 = \frac{\sigma_1 - \sigma_2}{2}$$



$$n_2 = n_3 = \pm \frac{1}{\sqrt{2}}$$

$$\tau_3 = \frac{\sigma_2 - \sigma_3}{2}$$

20.1.22

Octahedral Planes

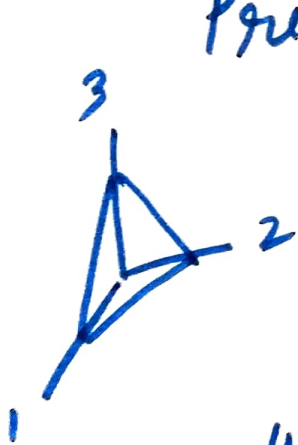
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Equally inclined
to all principal axes

$$n_1 = n_2 = n_3 = \frac{1}{\sqrt{3}}$$

Principal stresses



$\sigma_1, \sigma_2, \sigma_3$

Find
What are O.H. stresses
on the plane?

Find normal stress,
Shear stress
Total stress

$$S_i = \sigma_{ij} n_j$$

$$S_1 = \sigma_{11} n_1 + \sigma_{12} n_2 + \sigma_{13} n_3$$

$$S_2 = \sigma_{21} n_1 + \sigma_{22} n_2 + \sigma_{23} n_3$$

$$S_3 = \sigma_{31} n_1 + \sigma_{32} n_2 + \sigma_{33} n_3$$

$$\sigma_{ij} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix}$$

$$S_1 = \frac{\sigma_1}{\sqrt{3}}, \quad S_2 = \frac{\sigma_2}{\sqrt{3}}, \quad S_3 = \frac{\sigma_3}{\sqrt{3}}$$

$$S_n = S_1 n_1 + S_2 n_2 + S_3 n_3$$

$$S_n = \frac{1}{3} (\sigma_1 + \sigma_2 + \sigma_3)$$

$$S_{total}^2 = S_1^2 + S_2^2 + S_3^2$$

$$S_t^2 = \frac{\sigma_1^2}{3} + \frac{\sigma_2^2}{3} + \frac{\sigma_3^2}{3} \quad S_t^2 \checkmark$$

$$S_{total}^2 = S_n^2 + S_{shear}^2$$

$$S_{shear}^2 = S_{total}^2 - S_n^2$$

(2)

$$\sigma_{ij} = \begin{pmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{pmatrix}$$

$$= \begin{pmatrix} \sigma_m & 0 & 0 \\ 0 & \sigma_m & 0 \\ 0 & 0 & \sigma_m \end{pmatrix} + \begin{pmatrix} (\sigma_{11} - \sigma_m) & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & (\sigma_{22} - \sigma_m) & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & (\sigma_{33} - \sigma_m) \end{pmatrix}$$

Hydrostatic
Component

Stress
Deviator

$$\sigma_m = \frac{\sigma_{11} + \sigma_{22} + \sigma_{33}}{3} = \frac{I_1}{3}$$

Hydrostatic
Component



All principal stresses
are same

change axis \rightarrow get
Same thing

No off diagonal
elements

or
any plane stress
is only normal
stress

ex- Inside the sea



Stress Deviator (22)

State of pure
shear



\rightarrow Sum of diagonal
elements is zero

$$\rightarrow \sigma_1 + \sigma_2 + \sigma_3 = 0$$

Invariant

\rightarrow No normal
stress

\rightarrow Only shear
stress

\rightarrow needed for
plasticity
or deformation

Mohr's Circle

Graphical method to find out Principal planes & Principal stresses.

Two Dimension (2-D)

$$\begin{pmatrix} \sigma_{11} & \sigma_{12} & 0 \\ \sigma_{12} & \sigma_{22} & 0 \\ 0 & 0 & 0 \end{pmatrix}$$

plane stress

for shear stress

Sign Conventions

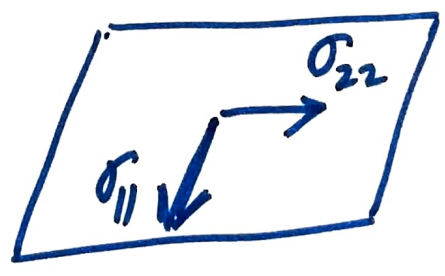
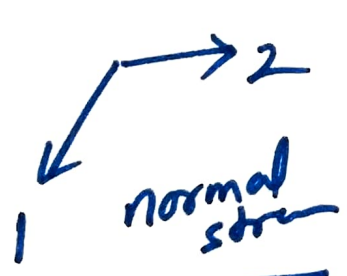


Clockwise

+ve

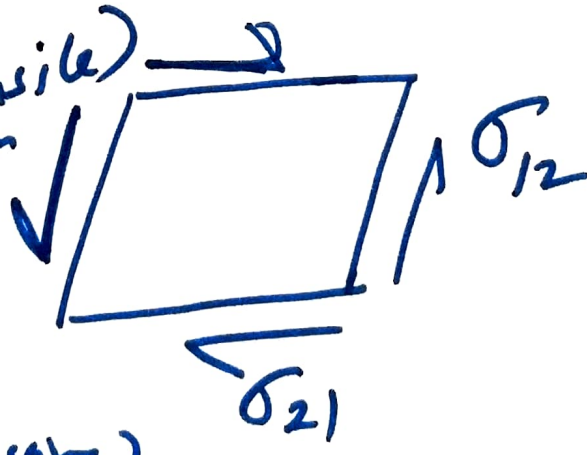
Anticlockwise

-ve



Pulling stress

+ve (tensile) stress

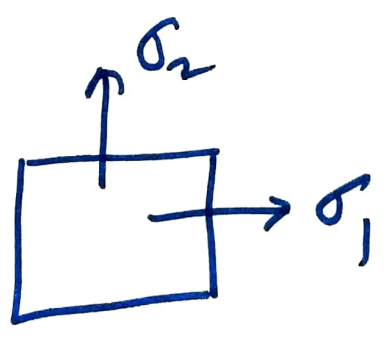
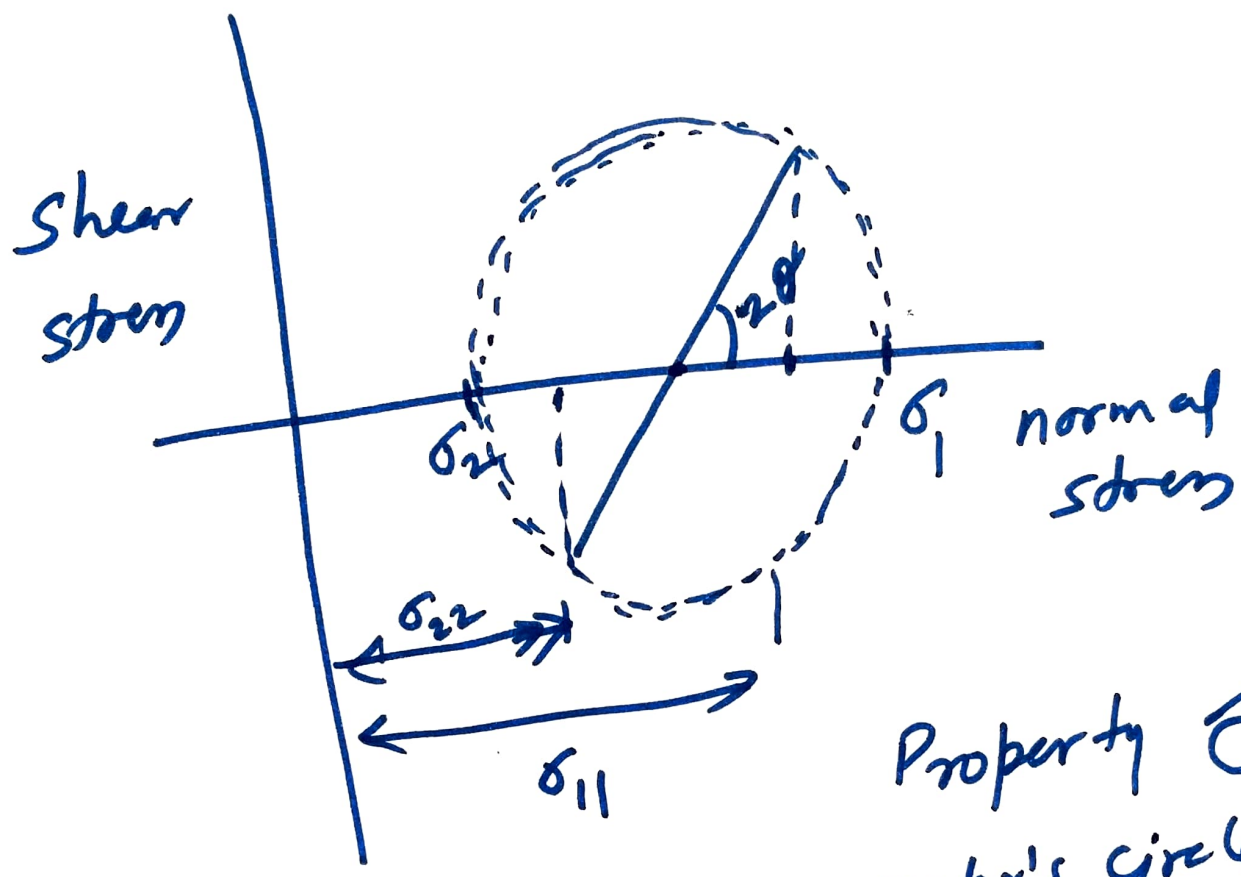


Pushing stress

-ve (Compressive) stress

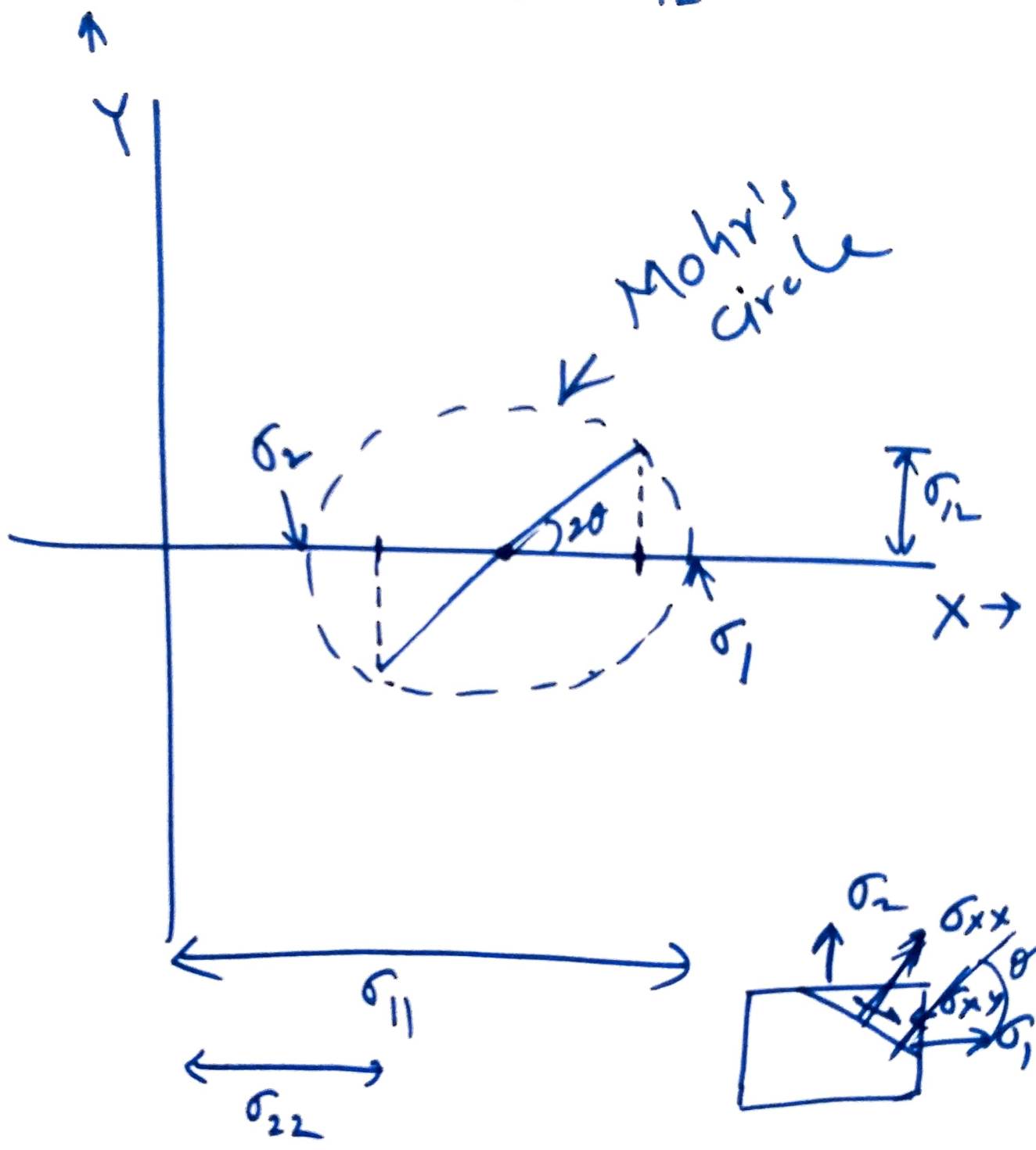
X axis \rightarrow normal stress

Y axis \rightarrow Shear stress



Property of Mohr's circle
 \Downarrow
Each point on circle represent stress on plane.

σ_{11} σ_{22} σ_{12}



$\sigma_1, \sigma_2 \Rightarrow$ Principal stress

$\sigma_1 > \sigma_2$

21.7.22

(26)

A rectangular sheet is loaded in plane stress to principal stresses of 100 MPa & 50 MPa.

Draw a Mohr's circle & find the values of the normal stress & shear stress on a plane with its normal inclined at an angle of 22.5 degrees to the major principal axis.

$$\sigma_1 = 100 \text{ MPa}$$

$$\sigma_2 = 50 \text{ MPa}$$

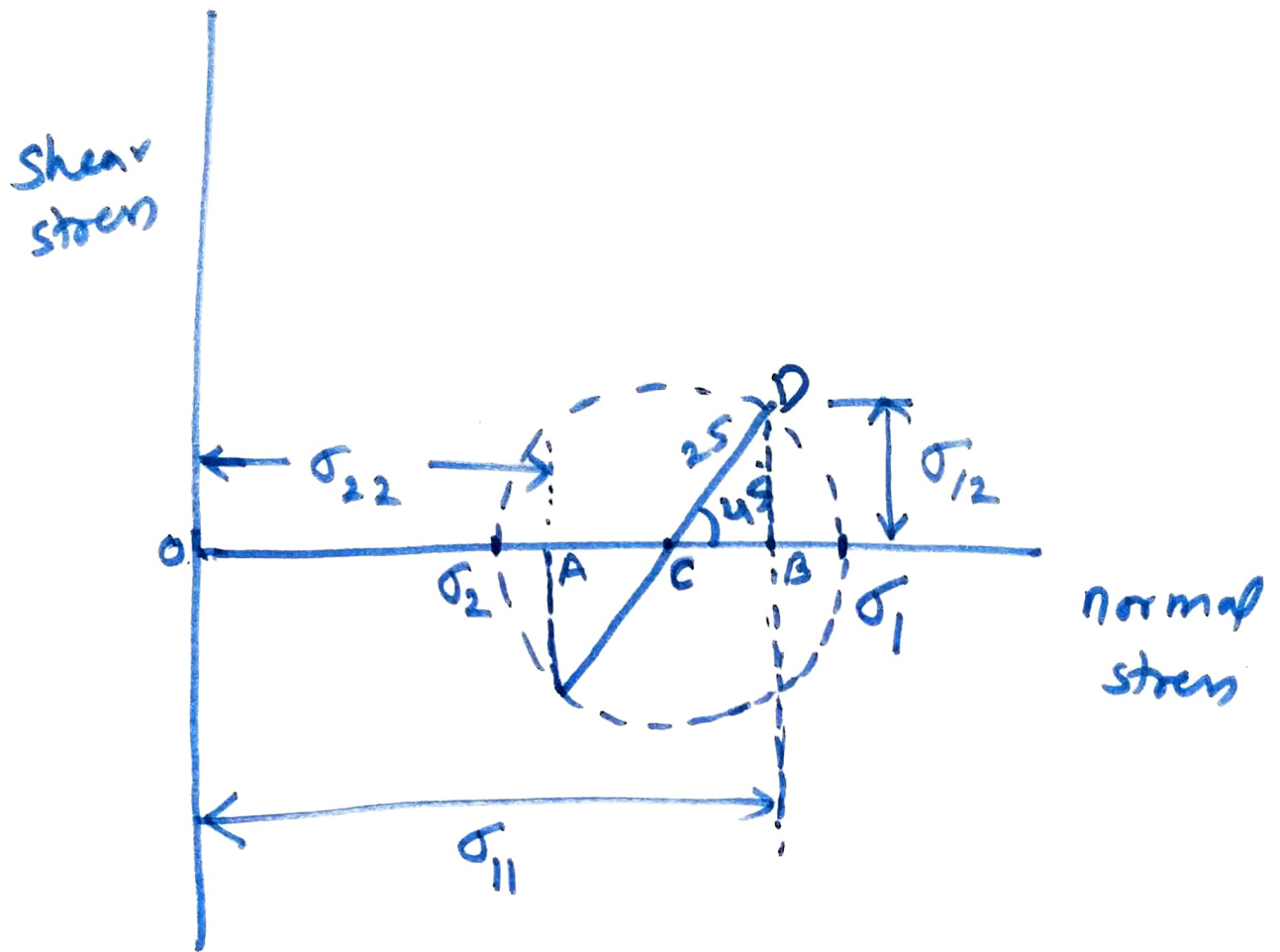
$$\left. \begin{array}{l} \sigma_{11} = ? \\ \sigma_{22} = ? \end{array} \right\} \text{normal stress}$$

$$\left. \begin{array}{l} \sigma_{12} = ? \end{array} \right\} \text{shear stress}$$

$$\theta = 22.5^\circ$$

Scale 1cm = 10 MPa

(27)



$$\begin{aligned}\sigma_{11} = OB &= OA + AC + CB \\ &= 50 + 25 + 25 \cos 45^\circ \\ &= 75 + 25 \times \frac{1}{\sqrt{2}}\end{aligned}$$

$$\begin{aligned}\sigma_{22} = OA &= OA \\ &= OA + AC \\ &= 50 + (25 - 25 \cos 45^\circ) \\ &= 50 + (25 - 25 \times \frac{1}{\sqrt{2}})\end{aligned}$$

$$\sigma_{12} = BD = 25 \sin 45^\circ$$

Find the principal stresses & 28
the orientation of the axes of
principal stress with the x, y axes
for the following situations.

Use Mohr's circle only.

(a) $\sigma_{11} = 5000 \text{ MPa}$

$$\sigma_{22} = 500 \text{ MPa}$$

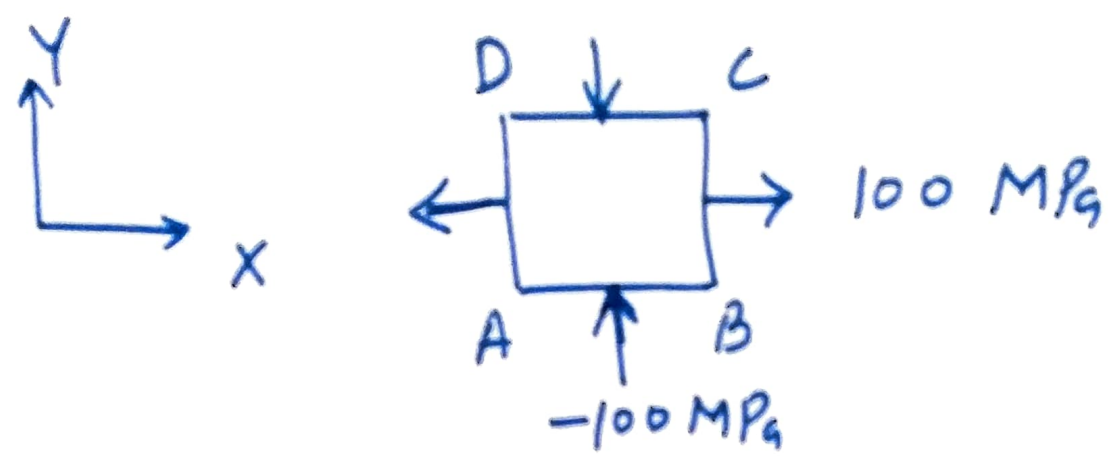
$$\sigma_{12} = -800 \text{ MPa}$$

(b) $\sigma_{11} = -6000 \text{ MPa}$

$$\sigma_{22} = 500 \text{ MPa}$$

$$\sigma_{12} = 2500 \text{ MPa}$$

(a) state of stress on an element of a thin plate is as follows



The edge AB is rotated counterclockwise by an angle 45°

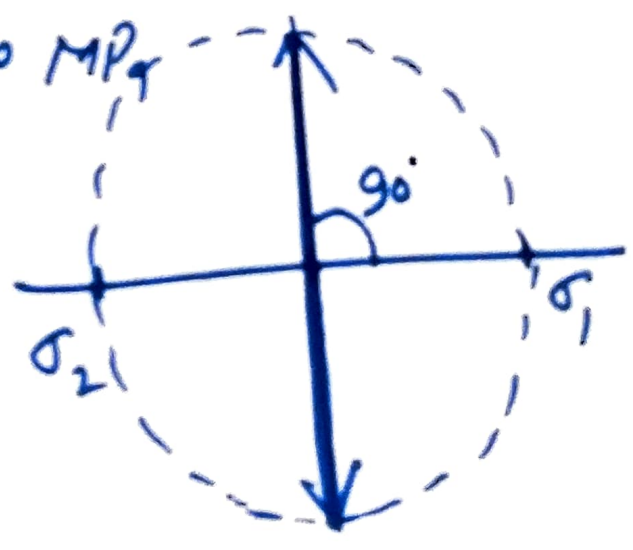
Sketch the state of stress after rotation. Use Mohr's circle.

$\sigma_1 = 100 \text{ MPa}$
 $\sigma_2 = -100 \text{ MPa}$

State of stress

Pure shear

$\tau = 100 \text{ MPa}$



Give the magnitude of principal stresses & direction of principal axes for the given two states of stress using Mohr's circle. (30)

$$(a) \quad \sigma_{11} = 15000 \text{ MPa}$$

$$\sigma_{22} = 4900 \text{ MPa}$$

$$\sigma_{12} = -6200 \text{ MPa}$$

(b)

$$\sigma_{11} = 12700 \text{ MPa}$$

$$\sigma_{22} = 7300 \text{ MPa}$$

$$\sigma_{12} = 7600 \text{ MPa}$$

At a point P in a body

$$\sigma_{xx} = 10,000 \text{ N cm}^{-2}$$

$$\sigma_{yy} = -5000 \text{ N cm}^{-2}$$

$$\sigma_{zz} = -5000 \text{ N cm}^{-2}$$

$$\sigma_{xy} = \sigma_{xz} = 10,000 \text{ N cm}^{-2}$$

$$\sigma_{yz} = 0$$

(a) Determine the normal & shear stresses on a plane that is equally inclined to all the three axes.

(b) Resolve this state of stress in hydrostatic & pure shear stress.

The stresses at a point, referred to the principal axes, are as follows

$$\begin{matrix}
 325 & 0 & 0 \\
 0 & 450 & 0 \\
 0 & 0 & 120
 \end{matrix}
 \text{ MPa}$$

What is the value of the max^m shear stress and on what plane(s) is it acting (show by a fig also).

$\sigma_1 > \sigma_2 > \sigma_3$

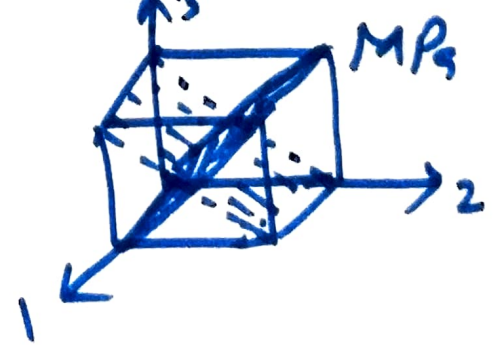
$\sigma_1 \rightarrow 450$

$\sigma_2 \rightarrow 325$

$\sigma_3 \rightarrow 120$

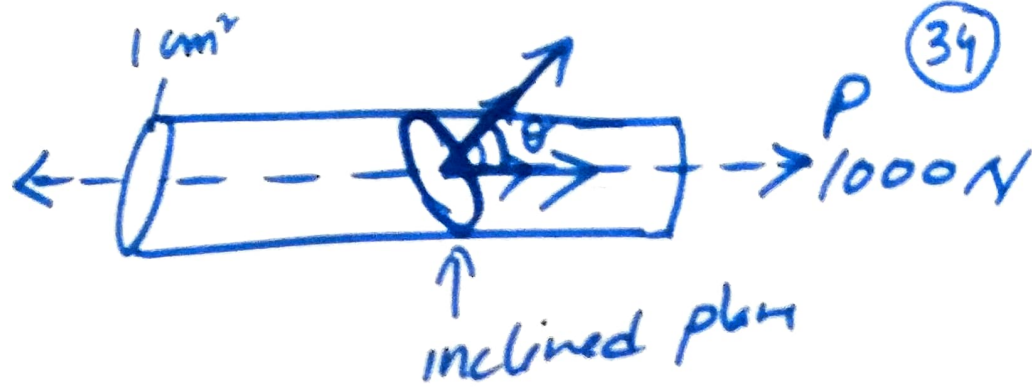
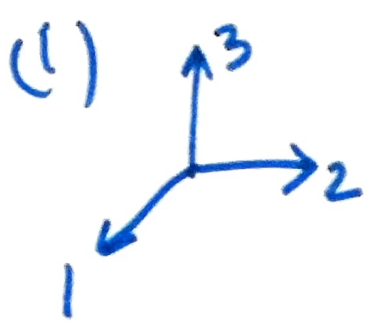
$$\begin{aligned}
 \tau_{max} &= \frac{\sigma_1 - \sigma_3}{2} \\
 &= \frac{450 - 120}{2}
 \end{aligned}$$

$\tau_{max} = 165 \text{ MPa}$



A load of 1000 N is applied along the axis of a circular bar of area 1 cm^2 . Find the normal stress and the shear stress on a plane inclined at an angle θ to the axis of the bar using the following methods

- (i) By resolution of the force along the given directions.
- (ii) By transformation rule for the stress tensor upon change of axes.
- (iii) By Drawing the Mohr's circle.



Area of inclined plane

$$= \frac{A}{\cos \theta}$$

$$\sigma = \frac{P}{A/\cos \theta} = \frac{P}{A} \cos \theta$$

Normal stress = $\sigma \cos \theta = \frac{P}{A} \cos^2 \theta$

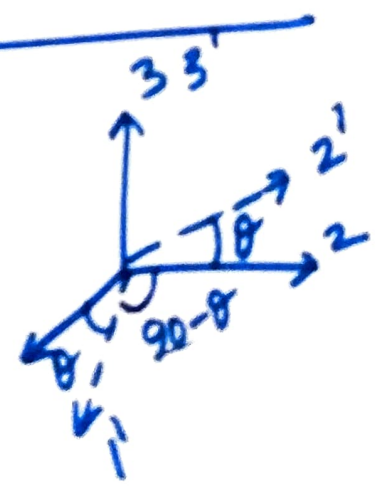
Shear stress = $\sigma \sin \theta$

$$= \frac{P}{A} \cos \theta \sin \theta$$

(ii)

$$\sigma'_{ij} = a_{ik} a_{jl} \sigma_{kl}$$

$$a_{ij} = \begin{pmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{pmatrix}$$



$$\sigma_{11}'$$

$$\sigma_{12}'$$

$$\sigma_{13}'$$

$$\sigma_{21}'$$

$$\sigma_{22}'$$

$$\sigma_{23}'$$

$$\sigma_{31}'$$

$$\sigma_{32}'$$

$$\sigma_{33}'$$

$$\sigma_{ij} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & \sigma_{22} & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$$\sigma_{ij}'^2 = \begin{bmatrix} & & \\ & & \\ & & \end{bmatrix}$$

$$a_{12} = \cos(90 - \theta) = \sin \theta$$

new / old

$$\sigma_{22} = \frac{P}{A} = \frac{1000}{1 \times 10^{-4}} = 10 \times 10^6 \text{ Pa}$$

$$= 10 \text{ MPa}$$

(iii)

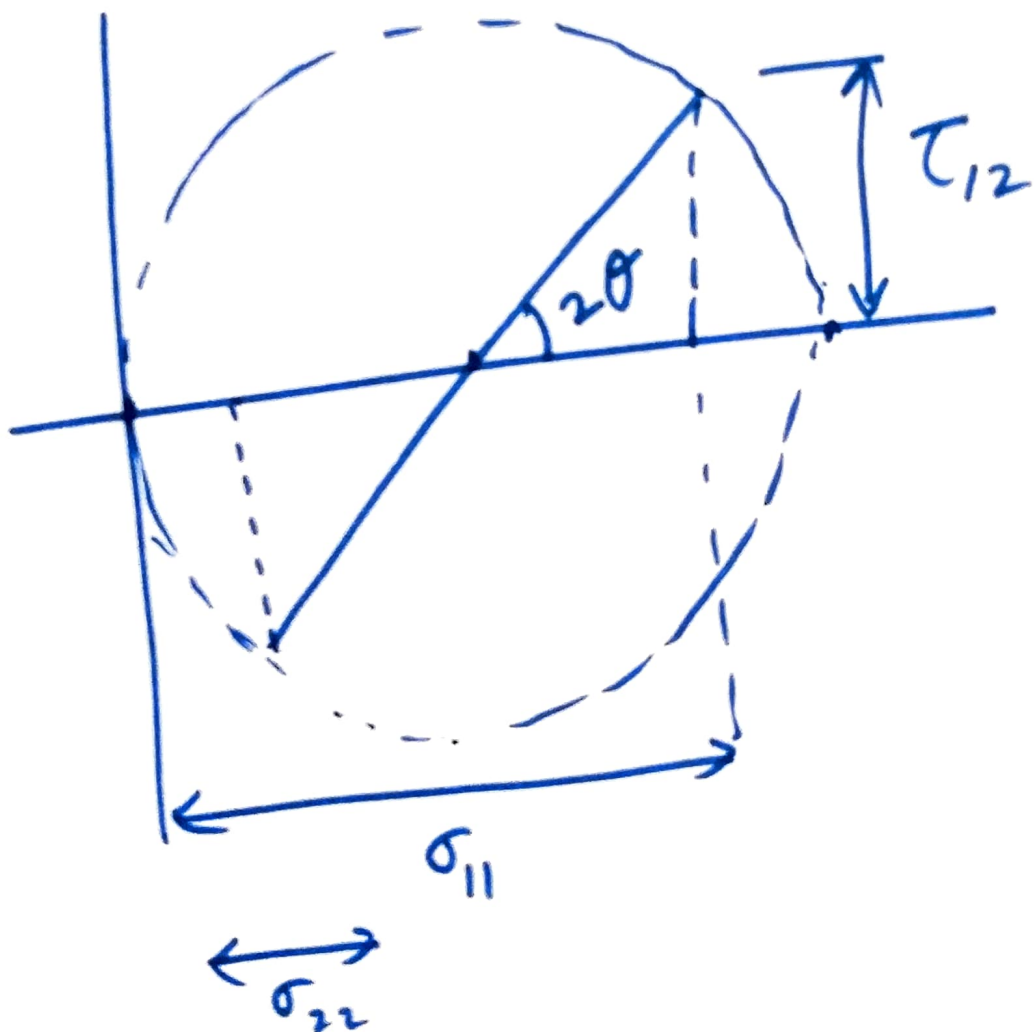
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$$\sigma_{22} = 10 \text{ MPa}$$

Principal stress

$$\sigma_2 = 10 \text{ MPa}$$

$$\sigma_1 = 0$$



State of stress at a point is given by (37)

$$\begin{pmatrix} 8000 & 2000 & -5000 \\ 2000 & -4000 & 3000 \\ -5000 & 3000 & 6000 \end{pmatrix} \text{ MPa}$$

Determine

(a) the total stress (Direction & magnitude) on plane described by direction cosine

$$n_x = \frac{1}{\sqrt{2}}, \quad n_y = \frac{1}{2},$$

$$n_z = -ve$$

(b) magnitude of normal & shear stresses on this plane.

$$n_1 = \frac{1}{\sqrt{2}} \quad n_2 = \frac{1}{2} \quad n_3 = -\frac{1}{2}$$

$$S_i = \sigma_{ij} n_j$$

S_1

S_2

S_3

$$S_{\text{total}} = \sqrt{S_1^2 + S_2^2 + S_3^2} = S$$

$$n_1' = \frac{S_1}{S}$$

$$n_2' = \frac{S_2}{S}$$

$$n_3' = \frac{S_3}{S}$$

(b)

$$S_{\text{normal}} = S_1 n_1 + S_2 n_2 + S_3 n_3$$

$$S_{\text{total}}^2 = S_{\text{normal}}^2 + S_{\text{shear}}^2$$

$$S_{\text{shear}}^2 = S_{\text{total}}^2 - S_{\text{normal}}^2$$

A new set of coordinate axes are obtained by rotating principal axes about the z axis by an angle θ . (39)

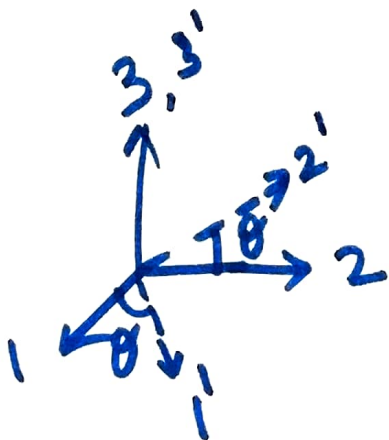
All the three principal stresses are equal. Show that with reference to the new axes:

(a) $\sigma_{11}' = \sigma_{22}' = \sigma_{33}' = \sigma_1 = \sigma_2 = \sigma_3$

(b) $\sigma_{ij}' = 0$ when $i \neq j$

(c) Are the new set of axes also principal axes?

What is this state of stress called?



$$a_{ij} = \begin{pmatrix} & & \\ & & \\ & & \end{pmatrix}$$

$$\sigma_{ij} = \begin{bmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{bmatrix}$$

$$\sigma'_{ij} = a_{ik} a_{jl} \sigma_{kl}$$

$$\sigma'_{11} =$$

$$\sigma'_{22} =$$

$$\sigma'_{33} =$$

$$\sigma'_{12} =$$

$$\sigma'_{13} =$$

$$\sigma'_{ij} =$$

$i \neq j$

Yes

Hydrostatic component.