

$$F_2 = -kx_1 + kx_2$$

$$\begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} = \begin{bmatrix} 2k & -k \\ -k & k \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix}$$

$$\{F\} = [k]_{2 \times 2} \{x\}$$

Multiplying both side by k^{-1}

$$[k^{-1}] \{F\} = [k^{-1}] [k]_{2 \times 2} \{x\}$$

$$[k^{-1}] \{F\} = I \{x\}$$

$\underbrace{\quad}_{\text{flexibility coefficient matrix}} \{a\}$

Physical Interpretation of stiffness coefficient -

$$[k^{-1}] \{F\} = \{x\}$$

$$\begin{bmatrix} a_{11} & a_{12} \\ a_{21} & a_{22} \end{bmatrix} = \begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} = \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix}$$

$$\Rightarrow a_{11}F_1 + a_{12}F_2 = x_1$$

$$\text{let } F_1 = 1 \text{ N}$$

$$F_2 = 0 \text{ N}$$

$$\Rightarrow a_{11} = x_1$$

Similarly,

$$\Rightarrow a_{21}F_1 + a_{22}F_2 = x_2$$

$$a_{21} = x_2$$

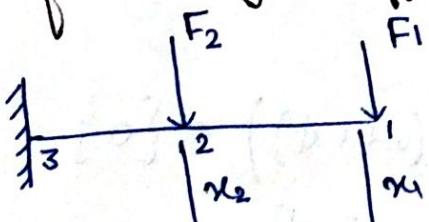
a_{11} → It is displacement of first node when a unit force applied on the first node and remaining force that acts on other node should be zero.

a_{21} → It is displacement of ^{first} second node when a unit force is applied on ~~second~~ ^{first} node and remaining force that acts on other node should be zero.

a_{12} → It is the displacement of ^{first} second node when a unit force is applied on the second node and remaining force that acts on the other node should be zero.

a_{22} → It is the displacement of second node when a unit force is applied on the second node and remaining force that acts on the other node should be zero.

flexibility coefficient in cantilever beam -



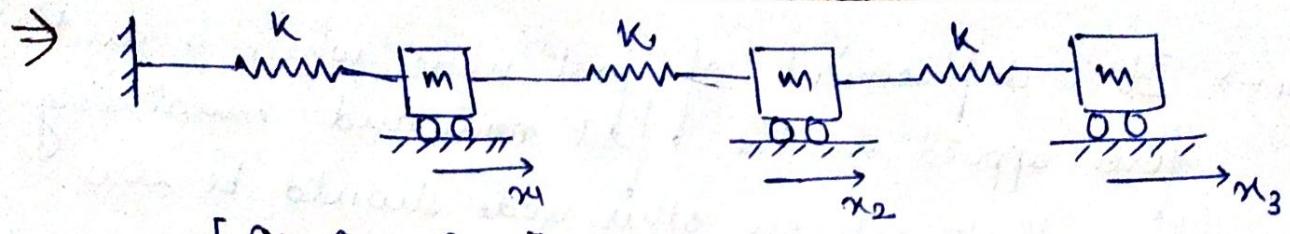
$$\begin{Bmatrix} F_1 \\ F_2 \end{Bmatrix} = \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{Bmatrix} x_1 \\ x_2 \end{Bmatrix}$$

$$\delta = P l^3 / 3 \epsilon I$$

$$\text{When } F_1 = 1 \text{ N, } F_2 = 0 \text{ N}$$

$$x_1 = a_{11} = l^3 / 3 \epsilon I$$

(Dunkerey method)



$$\begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{33} \end{bmatrix}$$

Apply unit force on block ①, ②, ③ simultaneously.

Duckerely method -

Duckerely method is an approximate method to find first natural frequency of a dynamical system.

$$\frac{1}{\omega_1^2} = \frac{1}{\omega_{11}^2} + \frac{1}{\omega_{22}^2} + \dots$$

~~Imp~~ Derivation -

$$[m] \{ \ddot{x} \}_{n \times n} + [k] \{ x \}_{n \times 1} = \{ 0 \}_{n \times 1}$$

$$\Rightarrow \{ \ddot{x} \}_{n \times 1} = \{ A \}_{n \times 1} \sin(\omega t + \phi)$$

$$\Rightarrow [-\omega^2 [m] + [k]] \{ A \}_{n \times 1} (\sin(\omega t + \phi)) = \{ 0 \}_{n \times 1}$$

Multiplying the above eqⁿ by $[k^{-1}]$ -

$$[-\omega^2 [k^{-1}] [m] + [I]] (\sin(\omega t + \phi)) = \{ 0 \}_{n \times 1}$$

$$[[a][m] - \frac{1}{\omega^2}[I]] (\sin(\omega t + \phi)) = \{ 0 \}_{n \times 1}$$

$$\det ([a][m] - \frac{1}{\omega^2}[I]) = 0$$

[do the above eqⁿ
was divided by ω^2
 $a/\omega = a$]

$$\Rightarrow \begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \\ a_{21} & a_{22} & \dots & \vdots \\ \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & \dots & a_{nn} \end{bmatrix} \begin{bmatrix} m_{11} & \dots & \vdots \\ \vdots & \ddots & \vdots \\ \vdots & \vdots & m_{nn} \end{bmatrix} - \begin{bmatrix} \frac{1}{\omega^2} \\ \vdots \\ \vdots \\ \frac{1}{\omega^2} \end{bmatrix}$$

$$\Rightarrow \begin{bmatrix} a_{11}m_{11} - \frac{1}{\omega^2} & a_{12}m_{22} & \dots \\ a_{21}m_{11} & a_{22}m_{22} - \frac{1}{\omega^2} & \dots \\ \vdots & \vdots & \ddots \\ a_{nn}m_{nn} - \frac{1}{\omega^2} & \dots & \dots \end{bmatrix}$$

$$\Rightarrow \left(\frac{1}{\omega^2} \right)^n - (a_{11}m_{11} + a_{22}m_{22} + \dots + a_{nn}m_{nn}) \left(\frac{1}{\omega^2} \right)^{n-1} + C_{n-2} \left(\frac{1}{\omega^2} \right)^{n-2}$$

$$\Rightarrow \left(\frac{1}{\omega^2} - \frac{1}{\omega_1^2} \right) \left(\frac{1}{\omega^2} - \frac{1}{\omega_2^2} \right) \dots \left(\frac{1}{\omega^2} - \frac{1}{\omega_n^2} \right) = 0$$

$$\Rightarrow \left(\frac{1}{\omega^2} \right)^n - \left(\frac{1}{\omega_1^2} + \frac{1}{\omega_2^2} + \dots + \frac{1}{\omega_n^2} \right) \left(\frac{1}{\omega^2} \right)^{n-1}$$

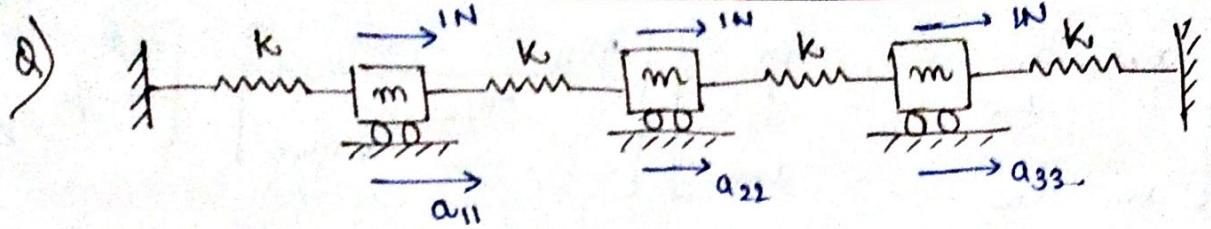
$$\Rightarrow \left(\frac{1}{\omega_1^2} + \frac{1}{\omega_2^2} + \dots \right)$$

$\frac{1}{\omega_1^2} \ggg \underbrace{\left(\frac{1}{\omega_2^2} + \frac{1}{\omega_3^2} + \dots + \frac{1}{\omega_n^2} \right)}$

This part is neglected.

$$\Rightarrow \frac{1}{\omega_1^2} = a_{11}m_{11} + a_{22}m_{22} + \dots + a_{nn}m_{nn}.$$

$$\Rightarrow \boxed{\frac{1}{\omega_1^2} = \sum_{i=1}^n a_{ii}m_{ii}}$$



$$\Rightarrow \quad \begin{array}{c} \text{Diagram of three blocks connected by springs with stiffness } k, \text{ each of mass } m, \text{ on a horizontal surface. A force of } 1N \text{ is applied to the first block from the left. The accelerations are labeled as } a_{11}, a_{22}, \text{ and } a_{33}. \\ \text{Equation: } F = k_{\text{eq}} \cdot x \\ \text{where } k_{\text{eq}} = k + \frac{k}{3} = \frac{4k}{3} \end{array}$$

$$\therefore F = k_{\text{eq}} \cdot x$$

$$1N = \frac{4k}{3} \cdot x$$

$$x = a_{11} = \frac{3}{4k}$$

When one mass of block is considered, other masses are considered to be zero.

\Rightarrow Now, On block 2 -

$$F = 1N$$



$$k_{\text{eq}} = \frac{k}{2} + \frac{k}{2} = k$$

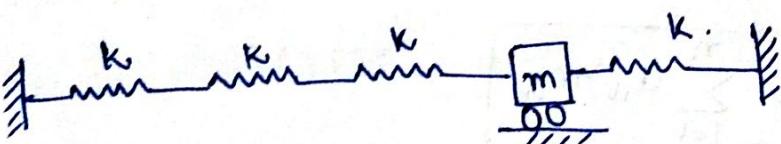
$$F = k_{\text{eq}} \cdot x \cdot x$$

$$1 = k \cdot x$$

$$x = a_{22} = 1/k$$

\Rightarrow Similarly, on block 3 -

$$F = 1N$$

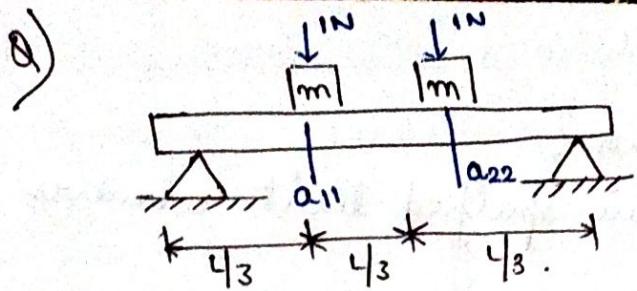


$$k_{\text{eq}} = k + \frac{k}{3} = \frac{4k}{3}$$

$$F = k_{\text{eq}} \cdot x$$

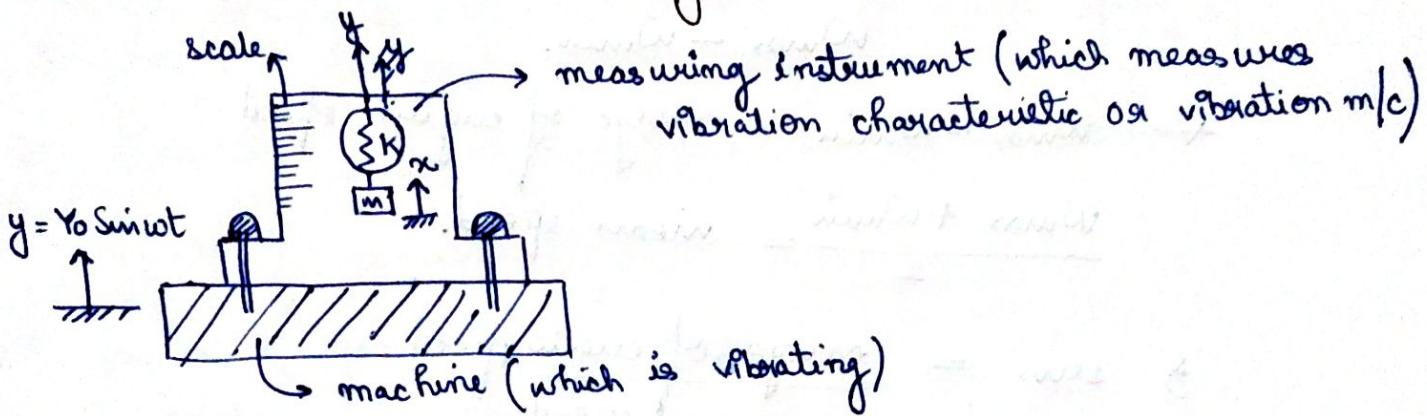
$$1 = \frac{4k}{3} \cdot x$$

$$x = a_{33} = 3/4k$$



$$\Rightarrow \frac{1}{\omega^2} = a_{11}m_{11} + a_{22}m_{22}$$

Vibration measuring instruments



$$a_{max} = -Y_0 \cdot \omega^2$$

If $x > y$, the spring will be compressed.



$$(x-y) = z$$

$$\ddot{x} - \ddot{y} = \ddot{z}$$

$$\ddot{x} = \ddot{z} + \ddot{y}$$

$$= \ddot{z} - Y_0 \omega^2 \sin \omega t$$

$$\sum F_x = ma$$

$$-k(x-y) = m\ddot{x}$$

$$m\ddot{x} + k(x-y) = 0$$

$$\text{Let } (x-y) = z$$

$$\Rightarrow m\ddot{x} + kz = 0$$

$$\Rightarrow m\ddot{z} + kz = m\omega^2 Y_0 \sin \omega t$$

$$z = z_0 \sin(\omega t + \phi)$$

$$\text{Where, } z_0 = \frac{m\omega^2 Y_0}{\sqrt{(k-m\omega^2)^2 + (c\omega)^2}}$$

- Article 16 Governor
- What is the function of governor?
Maintain speed of an engine within specified limit whenever there is variation of load.
- Distinguish b/w governor & flywheel.
- Insitiveness of governor -

$$\text{Sensitivity} = \frac{\omega_{\max} + \omega_{\min}}{2}$$

$$\frac{\omega_{\max} - \omega_{\min}}{\omega_{\max} + \omega_{\min}}$$

+ $\omega_{\max} - \omega_{\min} = \text{range of engine speed}$.

$\frac{\omega_{\max} + \omega_{\min}}{2} = \text{mean speed.}$

$\Rightarrow \text{sen.} = \frac{\text{range of engine speed}}{\text{mean speed.}}$
- A governor is said to be sensitive when it readily responds to a small change of speed. The movement of sleeve for a fractional change of speed is the measure of sensitivity.
- Hunting - Sensitivity of governor is a desired quantity, however if a governor is too sensitive, it may fluctuate continuously because when the load on the engine falls, the sleeve rises apparently to a maximum position, this shut off the fuel supply to the extent to effect falling speed.
- Isochronism - When the range of speed for governor is zero i.e. $\omega_{\max} = \omega_{\min}$.

- A governor with a range of speed zero is called isochronous governor. (sensitivity = ∞)

This means that for all position of sleeve or wall, the governor has the same speed.

- A pendulum-type governor can't possibly be isochronous.

→ Stability - A governor is said to be stable if it brings the speed of engine to the required value. The ball masses occupy a definite position for each speed of the engine within the working range.

* Stability & sensitiveness are two opposite terms.

- Effect of governor
- Power of governor
- Controlling force